# Equivalent Electrical Circuit Derivation of a 7MW Offshore Wind Turbine Gearbox

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#### Abstract

A thermal model of the main mechanical components of the Levenmouth Development Turbine powertrain has been developed in Matlab/Simulink in order to predict their efficiency curve as a function of torque and speed and their electrical analogues of current and frequency. A parametric loss estimation method has been adapted to the turbine gearbox architecture, its oil type and temperature and used to calculate the load and non-load dependent losses due to bearing, mesh, seal, churning and windage and pump losses. The gearbox losses have been expressed as notional equivalent electrical circuit parameters.

## 1 Introduction

The next generation of offshore wind turbines will have a wide range of powertrains: direct-drive through to multi-stage gearboxes and induction machines to permanent magnet generators (PMG). The ORECatapult (OREC) 7MW Levenmouth Development Turbine (LDT), is interesting because it has a gearbox and a medium speed PMG; this powertrain topology is becoming increasingly popular for offshore applications (e.g. Adwen, Vestas).

Modern wind turbines use torque control of the generator to track peak aerodynamic performance; the powertrain is designed to deliver this torque control with an appropriate balance of low cost and high efficiency. A significant number of in-service wind turbines operate in sub-optimal conditions [1] due to a number of reasons: component aging, uncertainty in aerodynamic and powertrain parameters, and the control and operational settings not being recalibrated in real time. The inability to recalibrate in real time to account for turbine parameter variations degrades the turbine efficiency, reducing capacity factor and increasing the associated cost of energy. A more dramatic decline is observed in offshore wind farms.

A limited number of studies have centred their attention on a detailed powertrain loss calculation. Basic approximations are typically used by researchers when the gearbox is considered. In [2], the authors state that the gearbox efficiency can vary between 95% and 98% depending on the number of stages and

the type of lubrication. In [3], Cotrell suggests that medium speed powertrains are more efficient than conventional high speed arrangements. A 1% per stage approximation due to churning is given for the intermediate and high speed stages. Mesh losses correspond to 0.5% per stage at rated power.

In [4], Polinder *et al* builds on the 1% per stage approximation carried out by Cotrell in [3] but also scales his results according to the rotational speed. Bywaters *et al* [5] and Li *et al* [6] estimated their gearbox losses making use of Cotrell's method. A more practical approach was developed by Chase *et al* [7] with the aim of testing the efficiency of high speed gearboxes. In [8], the authors analysed mechanical friction losses and windage losses as a function of the rotational speed and certain undefined parameters never disclosed to the public. In [9], Anderson and Loewenthal presented a detailed method to evaluate the efficiency of spur gears. Prakash del Valle in [10] and Duncan in [11] generated simple thermal models in Simulink of small gearboxes without cooling systems. The obtained results were compared with experimental data achieving a reasonable fit.

In industry, complex in-house pieces of software, based on finite element techniques, are produced to estimate the efficiency of the gearboxes [12].

Bearing this in mind, the authors have developed a detailed thermal model in Matlab/Simulink of the main mechanical components of the LDT powertrain to predict their efficiency curve as a function of torque and speed and their electrical analogues of current and frequency, as well as their temperatures. This will allow a comprehensive understanding of the system losses and help increase the capacity factor [13]. The parametric loss estimation method in [14] has been adapted to the LDT gearbox architecture, its oil type and temperature and used to calculate the load and non-load dependent losses due to bearing, mesh, seal, churning and windage and pump losses. The gearbox losses have been expressed as notional equivalent electrical circuit parameters, i.e. three electrical resistances, one dependent on torque, one dependent on rotational speed and one which depends on both. The equivalent electrical circuit reduces complexity and can be easily implemented in full wind turbine models.

# 2 Gearbox model development

In order to develop a parametric model of the gearbox, the guidelines given in the UK National Standard [14] have been

followed. The symbols and units, as well as the definition of the parameters used in this investigation to calculate the power losses are given in Table 1.

power loss	es are given in Table 1.		$M_1$ Bearing load dependent momen		nt Nm
Symbol	Meaning	Units	$M_2$	Cylindrical roller bearing axial load Dependent moment loss	Nm
$A_{\rm C}$	Gear casing surface area	$m^2$	$M_3$	Frictional moment of bearing seal	Nm
$A_{ m g}$	Arrangement constant for gearing	N/A	n	Rotational shaft speed	
a	Load modifying exponent	N/A		Pinion rotational speed	rpm
b	Diameter modifying exponent	N/A	$n_1$	•	rpm
$b_{ m w}$	Face width in contact with mating	mm	P	Bearing load	N
- "	element	element		Transmitted power	kW
$C_1$	Mesh coefficient of friction constant	N/A	$P_{ m B}$	Total bearing losses	kW
D	Outer diameter of gear element	mm	$P_{ m Bi}$	Individual bearing load power loss	kW
$d_{ m i}$	Bearing bore diameter	mm	$P_{ m L}$	Load dependent losses	kW
	-		$P_{ m M}$	Total gear mesh losses	kW
$d_{ m m}$	Bearing mean diameter	mm	$P_{ m Mi}$	Individual loaded mesh power loss	kW
$d_{ m o}$	Bearing outside diameter	mm	$P_{ m N}$	Non-load dependent losses	kW
$E_{ m p}$	Electric power consumed	kW	$P_{ m P}$	Total oil pump power required	kW
e	Bearing factor	N/A	$P_{\mathrm{Ps}}$	Shaft driven oil pump power	kW
$e_{ m p}$	Oil pump efficiency	N/A	$P_{\mathrm{S}}$	Total oil seal losses	kW
$F_{\mathrm{a}}$	Bearing axial load component	N	$P_{ m V}$	Heat generated	kW
$F_{ m r}$	Bearing radial load component	N	$P_{ m WB}$	bearing windage power	kW
$f_{ m m}$	Mesh coefficient of friction	N/A		loss	
$f_0$	Bearing dip factor	N/A	$P_{ m WBi}$	Individual bearing windage power loss	kW
$f_1$	Coefficient of friction for bearings	N/A	$P_1$	Bearing dynamic load	N
$f_2$	Cylindrical roller bearing factor	N/A	p	Operating oil pressure	N/mm <sup>2</sup>
$f_3$	Bearing seal factor	N/A	Q	Oil volumetric flow	l/min
$f_4$	Bearing seal factor	N/A	$r_{ m o1}$	Pinion outside radius	mm
g	Load intensity modifying exponent	N/A	$r_{o2}$	Gear outside radius	mm
$H_{\rm s}$	Sliding ratio at start of approach	N/A	$r_{ m w1}$	Pinion operating pitch radius	mm
$H_{\rm t}$	Sliding ratio at end of recess	N/A		Gear operating pitch radius	
h	Pitch line velocity modifying	N/A	$r_{ m w1}$ $T_1$	Torque on the pinion	mm Nm
j	Viscosity modifying exponent	N/A		Gear ratio	N/A
K	Load intensity	N/mm <sup>2</sup>	u V	Pitch line velocity	
M	Mesh mechanical advantage	N/A	V	r nen mie veloeity	m/s

 $M_0$ 

No-load torque moment on bearings

Nm

<i>Y</i> , <i>Y</i> <sub>2</sub>	Bearing factors	N/A
<i>Z</i> 1	Number of pinion teeth	N/A
<i>Z</i> 2	Number of gear teeth	N/A
$a_{ m w}$	Operating transverse pressure angle	degrees
$oldsymbol{eta}_{ m w}$	Helix angle	degrees
η	Efficiency	%
v	Kinematic viscosity of oil at room temperature	cSt

Table 1: Symbols and units [14]

By using a model like this, the user can easily estimate the power losses of a particular gearbox with high level of accuracy due to its parametric nature. The three stage epicyclic planetary gearbox of the LDT wind turbine has been considered in this study. Figure 1 shows a sketch of the gearbox arrangement.

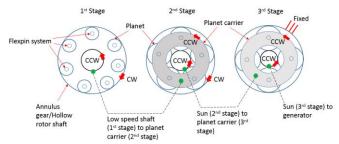


Figure 1: LDT wind turbine gearbox arrangement

As seen, a hollow rotor shaft rotating clockwise acts as a gearbox housing and as the annulus gear for stages 1 and 2. Seven planet gears are mounted onto flexpin systems in the first stage. Planets do not orbit around the sun gear. The flexpins are attached to a back plate, which is bolted to the outer casing. The low speed shaft in the first stage rotates anticlockwise with the same speed as the wind turbine rotor and drives the planet carrier of the second stage. In the second stage there are 5 planet gears mounted onto flexpins. The sun gear, in this case, rotates anticlockwise again and at higher speed than if the annulus would be fixed and drives the planet carrier of the third stage. This second stage creates a superposition. This concept is beneficial in order to achieve high gear ratios. The third stage is rather conventional with a stationary annulus gear and four planets mounted onto flexpins and with the sun gear rotating anticlockwise. The sun gear in the third stage is connected to the electrical generator. The overall gear ratio is 10.7. Further details on the turbine and gears characteristics can be found in Tables 2 and 3, respectively. An oil pump lubrication system driven by one of the reducer shafts was considered. The gearbox was assumed to be fully drained off.

So as to develop a loss model for this machine the guidelines given in [14] have been followed. The thermal rating calculation is based on the fact that the heat generated in a gear

drive  $P_V$  is equal to the heat dissipation of the gear drive,  $P_Q$ . Thus  $P_Q = P_V$ . The heat generated in a gear drive,  $P_V$ , can be estimated by adding up load dependent losses,  $P_L$ , and non-load dependent losses,  $P_N$ .

$$P_{\rm V} = P_{\rm L} + P_{\rm N} \,. \tag{1}$$

The load dependent losses are a function of the input power,  $P_A$ ,

$$P_{\rm L} = f(P_{\rm A}). \tag{2}$$

Rearranging terms in Equation (1), the basic heat balance equation can be found,

$$P_{\rm O} - P_{\rm N} - f(P_{\rm A}) = 0. (3)$$

When Equation (3) is satisfied, the overall efficiency can be estimated as follows,

$$\eta = 100 - \frac{P_{\rm L} + P_{\rm N}}{P_{\rm A}} \times 100.$$
(4)

The load dependent losses,  $P_L$ , can be calculated using Equation (5),

$$P_{\rm L} = \sum P_{\rm B} + \sum P_{\rm M} \tag{5}$$

where  $P_{\rm B}$  and  $P_{\rm M}$  are the bearing load losses and the mesh losses of the gears, respectively. The non-load dependent losses can be found by summing up the seal losses,  $P_{\rm S}$ , the bearing windage losses,  $P_{\rm WB}$ , and the pump losses,  $P_{\rm P}$ .

$$P_{\rm N} = \sum P_{\rm S} + \sum P_{\rm WB} + \sum P_{\rm P}. \tag{6}$$

In order to calculate the gear tooth mesh power losses of an individual gear Equation (7) was utilised. For information about symbols and their units see Table 1.

$$P_{\rm Mi} = \frac{(f_{\rm m} T_1 n_1 cos^2 \beta_{\rm w})}{9549M} \tag{7}$$

The equation to calculate the mechanical advantage, M, is as follows,

$$M = \frac{\left(2\cos\alpha_{\rm w}(H_{\rm s} + H_{\rm t})\right)}{H_{\rm s}^2 + H_{\rm t}^2} \tag{8}$$

where the sliding ratio at start of approach,  $H_s$ , is calculated using Equation (9),

$$H_{\rm s} = (u+1) \left[ \left( \frac{r_{\rm o2}^2}{r_{\rm w2}^2} - \cos^2 \alpha_{\rm w} \right)^{0.5} - \sin \alpha_{\rm w} \right] \tag{9}$$

and the sliding ratio at the end of recess,  $H_t$ , is calculated using Equation (10),

$$H_{\rm t} = \left(\frac{u+1}{u}\right) \left[ \left(\frac{r_{01}^2}{r_{\rm w1}^2} - \cos^2 \alpha_{\rm w}\right)^{0.5} - \sin \alpha_{\rm w} \right]. \tag{10}$$

Finally the load intensity, K, is given by Equation (11)

$$K = \frac{1000T_1(z_1 + z_2)}{2b_{\rm w}(r_{\rm w1})^2 z_2}$$
 (11)

where  $f_{\rm m}$ , coefficient of friction of the gears was found with Equation (12).

$$f_{\rm m} = \frac{v^j K^g}{C_1 V^h} \tag{12}$$

For the calculation of the bearing load dependent losses, the load dependent torque,  $M_1$ , on each bearing as a function of the applied load is equal to

$$M_1 = \frac{f_1(P_1)^a (d_{\rm m})^b}{1000} \tag{13}$$

whereas the axial dependent moment is equal to

$$M_2 = \frac{f_2 F_a d_{\rm m}}{1000}. (14)$$

The load dependent power loss for an individual bearing can be calculated using Equation (15).

$$P_{\rm Bi} = \frac{(M_1 + M_2)n}{9549} \tag{15}$$

In order to estimate the bearing non-load dependent losses, the frictional moment,  $M_0$ , is characterised by two equations depending on the value of the product of the oil viscosity and the rotational shaft speed.

If *vn* < 2000:

$$M_0 = 1.6 \times 10^{-8} f_0 d_{\rm m}^3 \,. \tag{16}$$

If  $vn \ge 2000$ :

$$M_0 = 10^{-10} f_0(vn)^{2/3} d_{\rm m}^3 \,. \tag{17}$$

The frictional moment,  $M_3$ , of a bearing sealed at both ends is given by Equation (18)

$$M_3 = \frac{\left(\left(\frac{d_{\rm m}}{f_3}\right)^2 + f_4\right)}{1000}.$$
 (18)

The windage power loss for an individual bearing can be calculated using Equation (19).

$$P_{\text{WBi}} = \frac{(M_0 + M_3)n}{9549} \tag{19}$$

The oil seal power losses were estimated from Equation (20).

$$P_{\rm S} = \frac{T_{\rm S}n}{9549} \tag{20}$$

where  $T_{\rm S} = 3.737 \times 10^{-3} D_{\rm s}$ , with  $D_{\rm s}$  being the diameter of the shaft.

The last type of loss considered was the lubrication oil pump losses,  $P_{Ps}$ , which was estimated as follows,

$$P_{\rm Ps} = \frac{Qp}{60e_{\rm p}} \tag{21}$$

## 2.1 Modelling of external conditions

The loss model was adapted to the characteristics of the LDT wind turbine. The turbine specifications are listed in Table 2. The gearbox input parameters were estimated according to the results achieved from a wind field simulation in QBlade [16]. A 60 seconds simulation was set up considering the rotor radius, hub height, rated wind speed as the mean wind speed, measurement height equal to the hub height, a turbulence intensity of 10% and a roughness length of 0.01m so that the shear layer effect can be included in the study. 60 time steps were assumed (1 per second) with 20 points per direction. Points wind speeds were averaged obtaining a single value for the wind speed in each time step. Then, the rotor speed was found by interpolating between the values given in Table 2, considering that the rotor rotates at 5.9rpm and 10.6rpm when the wind speed is 3.5m/s and 25m/s, respectively. The power corresponding to each time step was acquired by extrapolation, considering the rated power and the rated wind speed. Intermediate and high speed shafts rotational speeds were calculated considering the wind turbine rotor speed and the corresponding gear ratios.

Rated power	7MW	
Rotor diameter	171.2m	
Hub height	110.6m	
Rated wind speed	11.5m/s	
Wind speed	3.5 - 25 m/s	
Rotor speed	5.9 – 10.6rpm	
Powertrain	Medium speed (400rpm)	
Overall gear ratio	10.7	

Table 2: Wind turbine specifications [15]

## 2.2 Thermal model development in Matlab/Simulink

The development of the complete thermal model of the LDT gearbox implied the creation of a code in Matlab [17], which governs the behaviour of the entire model in Simulink and allows the user to take into consideration distinct features, such as oil degradation with temperature, forces acting on helical gears or to implement a cooling/heating system. The fundamental characteristics of the gearbox components, such as gears or bearings mass, surface area, thermal conductivity, specific heat and density, have been estimated using the SolidWorks (ISO standard) toolbox [18]. All the gears were CAD generated using the data provided in Table 2 so that the unknown characteristics could be found. Shafts, planet carriers and bearings were also CAD generated using the said toolbox, although the data used were appropriate approximations made by looking at the dimensions of the gears and their rotational speeds.

Gear	No. teeth	Mate teeth	External
element			diameter (mm)
Annulus gear (1 <sup>st</sup> )	260	52	3000
Planet gear (1st)	52	260	805
Annulus gear (2 <sup>nd</sup> )	260	120	3000
Planet gear (2 <sup>nd</sup> )	120	260/20	500
Sun gear (2 <sup>nd</sup> )	20	120	220
Annulus gear (3 <sup>rd</sup> )	260	123	3000
Planet gear (3 <sup>rd</sup> )	123	260/14	1250
Sun gear (3 <sup>rd</sup> )	14	123	160

Table 3: Gear specifications

Double-row cylindrical roller bearings with full complement were assumed for the main bearings (low speed, intermediate speed and high speed), as well as for the flexpin systems. So as to consider the oil degradation, oil viscosity variation with temperature data, as given in [19], were plotted and a trend line fitted. The result obtained is well represented by Equation (22)

$$v = 2785 \exp(-0.045 T_{\text{sump}})$$
 (22)

where  $T_{\text{sump}}$  is the temperature of the oil. For the creation of the model in Simulink [17], thermal mass blocks were necessary to represent the gearbox elements (gears, bearings, shafts, oil reservoir, air cavity and outer casing). According to the type of heat transfer, conduction (solid-solid), convection (solid-fluid) or radiation (solid-fluid) blocks have been used. These are joint to the corresponding gearbox components simulating the transfer of heat. The changing power losses, are also connected to the thermal masses through ideal heat source blocks. Sensors are placed all over the gearbox to measure the temperatures of every component. The atmosphere has been simulated by using an ideal temperature source block with a constant input temperature of 294K.

The oil cooling/heating system, which starts working when the oil temperature reaches a designated value, was generated using ideal heat source blocks governed by Equations (23) and (24)

$$heating_{gain} = n \times Q_{heating} \times \rho_{water} \times C_{pwater}$$

$$cooling_{gain} = n \times Q_{cooling} \times \rho_{air} \times C_{pair}$$
(23)

where  $Q_{\text{heating}}$  is the water volumetric flow,  $\rho_{\text{water}}$  corresponds to the water density and  $C_{\text{pwater}}$  to water's heat coefficient, whereas,  $Q_{\text{cooling}}$  is the air volumetric flow,  $\rho_{\text{air}}$  corresponds to the air density and  $C_{\text{pair}}$  to air's heat coefficient. The results for the total power losses and the oil temperatures with and without cooling systems are depicted in Figure 2.

As observed, the total power losses are maximum at 32 seconds, coinciding with the maximum wind speed (11.98m/s). This represents a loss of 1.3% at this particular moment, with the non-load dependent losses accounting for 0.35% and the load dependent losses for 0.95%. From Figure 2(b)(c), it can be understood how the model accurately simulates the effect of the cooling system, as a smooth control of the temperature is achieved.

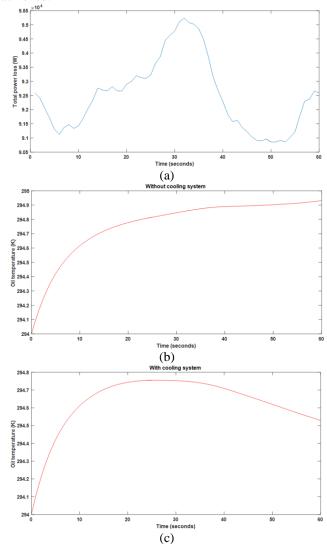


Figure 2: (a) Total power loss " $P_V$ "; (b) Oil temperature vs. time without cooling system; (c) Oil temperature vs. time with cooling system

## 2.3 Gearbox equivalent electrical circuit derivation

Once the thermal model was completed, the results retrieved from the simulation were analysed and depending on the influence of each variable (in this case torque and rotational speed are of interest) on the power losses, the power losses were classified into three different categories: torque dependent, rotational speed dependent and torque and rotational speed dependent. By knowing the power losses and the current flowing, electrical resistances can be estimated (R=  $P/I^2$ ), e.g.  $R_1$  (torque dependent), and introduced. For the rotational speed dependent losses, the resistance can be directly added to the variable resistance  $R_2$ /s, which depends on a slip, which can be defined in terms of the turbine rotor speed, and hence upon the speed of the generator. In the case of the power losses dependent of both the torque and the rotational speed these can be approximated and separated into components  $R_1$ ,  $R_2$  and  $R_3$ . See Figure 4.

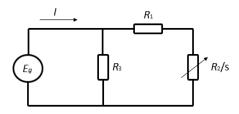


Figure 4: Equivalent electrical circuit of the gearbox

#### 3 Discussion and conclusions

The thermal model development of the LDT gearbox will help understand the system losses. By considering the thermal behaviour of the powertrain mechanical components, the turbine control strategy can be adapted in order to maximize the energy capture. In addition, with the creation of the equivalent electrical circuit of the gearbox the overall complexity of the system is considerably reduced and full wind turbine studies could be carried out by introducing a simplified generator and converter equivalent electrical circuits.

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